

Analysis of Fly Ash as Filler on Glass Fiber Reinforced Epoxy Composites

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ABSTRACT: Fly ash, a waste by-product is generated abundantly by combustion of coal in thermal power plant. The percentage variation of fly ash plays an important role in the enhancement of mechanical properties of glass fiber reinforced epoxy composite. The present study deals with the effect of percentage variation of filler contents on composites. As per ASTM D 3039 the CAD model is designed and three different specimens of various filler contents (10%, 30%, and 40%) and one pure composite specimen were numerically analyzed on ANSYS 16.0. It was found that the pure composite material enhanced the most strength.

Keywords: Epoxy Composite, Fly ash filler material, FEM, UTM Experimentation

I. INTRODUCTION

Nowadays composite materials are widely used in automotive, aerospace industries. Composite materials are lighter in weight compared to metals and woods. Their lightness is important in various industries for example less weight means better fuel efficiency. Glass fiber reinforced epoxy composites are results in an attractive combination of physical and mechanical properties which cannot be obtained by monolithic materials.

Epoxy resins are widely used as a matrix in many fiber reinforced composites; they are a class of thermoset materials of particular interest to structural engineers owing to the fact that they provide a unique balance of chemical and mechanical properties combined with wide process ing versatility. Within reinforcing materials, glass fibers are the most frequently used in structural constructions because of their specific strength properties. The ease of availability of glass fiber and economic manufacturing techniques adopted for production of components. Developments are still under to increase their properties. One of the methods to increase the strength of glass fiber reinforced epoxy composites is to add various filler materials. These filler materials act as additional reinforcing components and show their mechanical properties. Among various filler materials fly ash is one of the filler material that can be used as filler on composites to enhance their mechanical properties.

Fly ash is a waste by-product which is generated by combustion of coal in thermal power plants. Increasing production of fly ash year by year, from the coal based thermal power plant, it affects serious problem in terms of safe disposal and

utilization. The utilization of fly ash as filler materials are generally acts as inert materials which are used in composites to reduce the material cost, to improve the mechanical properties to some extent and in some cases to improve process ability. Reduction in filler size gives better enhancement in properties due to uniform distribution of particles in polymer matrix. As per ASTM C618 fly ash has been classified into two categories, Class F and Class C. Generally it is a mixture of oxides rich in silicon (SiO_2), iron (Fe_2O_3) and aluminum (Al_2O_3). It depends upon the source of coal; which contains different properties of silica, alumina, oxides of iron, calcium, magnesium and other elements like C, Ti, Mg, etc. So that the fly ash has properties combined of spherical particles and that of metals and metal oxides.

II. LITERATURE SURVEY

Baljeet Kumar, Rajeev Garg and Upinderpal Singh have studied the utilization of fly ash as filler material in polymer composites is considered important from both economic and commercial point of view. Fly ash is used as reinforcing filler in High density polyethylene (HDPE) to develop lightweight composites. After surveying they have concluded that fly ash is used as reinforcing filler material in High density polyethylene (HDPE) some studies have pointed to the excellent compatibility between fly ash and polymer. Modification of Fly ash accompanied by compatibilization leads to the substantial improvement properties of the composites. However, it is obvious that the potential as reinforcing fillers in polymers especially for Fly ash/HDPE composites have not been fully brought into play. [1]

K.ThomasPaul,S.K.Sathpathy,IManna,K.K.Chakra borty,

G.B. Nando et al. studied the size reduction of fly ash frommicrometer level to nano level which is achieved by highenergyballmilling.Theaverageparticlesizehasbeenreduced from 60µm to 1480 nm, a reduction of nearly 405timesinmagnitude.[2]

S.R. Chauhan, Anoop Kumar, I Singh and Prashant Kumar etal. studied the design and experiment method which can beused to analyze the coefficient of friction and the dry slidingwear of polymer matrix composites and found the results,coefficient of friction decreases with the addition of 10 wt %to 20 wt % of fly ash and wear resistance is increased for theadditionof10wt %to20wt%offlyash. [3]

R.Satheesh Raja, K.Manisekar, V.Manikandan et al. studiedthe effect of fly ash filler size on mechanical properties ofpolymermatrixcomposites.Thecompositespecimensareprepared in the four different sizes (50 µm, 480 nm, 350 nm,and300nm)offlyashfiller materialsbyusingCAD molding process. Mechanical testing such as hardness andimpact testwas carriedout.It is found that the 300 nm sizefly ash filler impregnated polymer compositeyields betterimpact energy (14 J) and hardness value (35 Hv) than others.Thusbydecreasingthesizeofflyashfillerleadsto increase the interference bond between the polymeric matrix and thesolid fillers.[4]

JitendraGummadi,G.VijayKumar,GuntiRajeshhave preparedthe five different particlesizes of fly ash are usedforsamplepreparation.Percentagevariationoffly ashinpolypropyleneis0,10,15,20,25.Thecompositestspecimens are prepared using injection molding machine withhandlayuptechniqueasperASTMD3641standards.Bendingtestsonthespecimensarecarriedoutbyusingtensometer.ModulusandFlexuralstrengtharecalculatedfrom the obtained load values and the result is analyzed forthepreparedsamples.WithflyashaddedtothePolyp

ropyleneimprovesflexuralmodulusandflexuralstrength ,but dramatically decreases percentage elongation atbreak. Finest particles showed best flexural strength at allconcentrations.[5]

PatilDeogonda, Vijaykumar N Chalwa et al. have used TiO₂andZnS as fillermaterial on GFRP laminated compositesandevaluatetheTensile,BendingandImpactstrengthincreaseswithadditionoffillermaterial.Zn Sfilledcomposite shows significantly good results than TiO₂ filledcomposites. ZnS filled composite shows more tensile load incomparisonwithunfilledandTiO₂filledcomposites. TiO₂andZnSfillermaterialmakesmaterialharderandbrittlewhichisthereasonforreductioninimpacttoughnessvalue.[6]

S.D.Saravanan,M.SenthilKumaretal.havestudiedthe effect of mechanical properties on rice husk ash reinforcedAluminum alloy (AlSi10Mg) matrix composites. A rice huskash particle of 3,6,9 & 12% by weight were used to developmetal matrix composites using liquid metallurgy route andfoundtheresultsthatthetensilestrength,compressionstrength,andhardnessincreaseswithincreaseinweightfraction and ductility get decreases with increase in weightfractionofricehuskash.[7]

From the above literature survey it is clear that, the tensiletests were conducted for TiO₂, ZnS and rice husk ash as fillermaterial on composites and some flexural, impact test wereconducted for fly ash as filler material with the different sizeof fly ash. So in this project, different composite specimenswill be prepared by varying percentage of fly ash& carry outtensile testonUTM.

III. DESIGN AND ANALYSIS

A. MaterialSpecification:

The materials used to prepare the specimen are E-Glassfiber,Epoxyresin(LY556),Hardener(HY951)
 Material: Epoxy, Glass Fiber, Fly Ash
 Young's Modulus:5000 -35000MPa
 Poisson's Ratio:0.24-0.4
 Density:1800-1850kg/m³

Table1: Filler Material Specimen Details

Sr. No.	E-Glassmat		
	Glassfibercontent %	Epoxy	Fillercontent in% (flyash)
1	60	40	-
2	50	40	10
3	30	40	30

4	20	40	40
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Table2: Testspecimendetail

Testspecimens	ASTM	Size
Tensiletestspecimen	D-3039	250x25x2.5mm.

B. CADMODEL:

To prepare the CAD model of specimen ANSYS 16.0 Designmodelerisused.

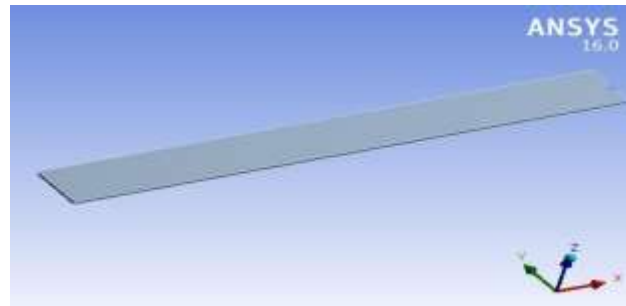


Figure1: CAD Model ofspecimen

C. DiscretizationorMeshing:

For the discretization of model a hexahedron element with standard program controlled mesh is used and having the number of nodes 2640 and number of elements 1890.

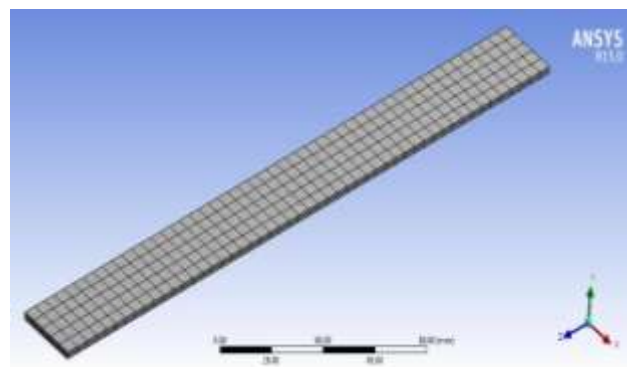


Figure2: Discretizationofspecimen

D. BoundaryCondition&Loading:

To apply tension on specimen one end is made to fix and another end is applied with tension load.

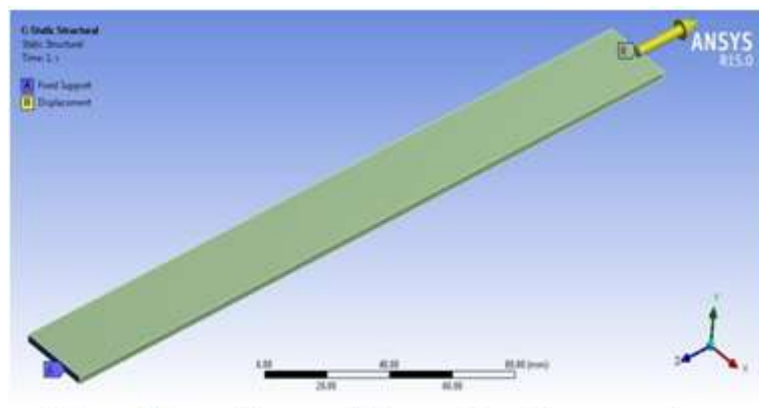


Figure3:Boundary conditions and loading on specimen

E. Orientations:

Given orientations are on X-axis=Normal directions, Y-axis
=Transverse directions, Z-axis=Fiber directions.

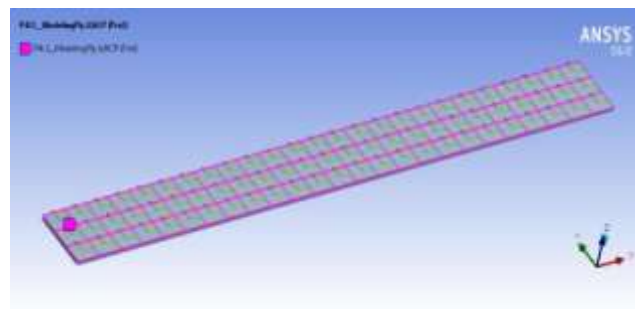


Figure4:Orientations of specimen

F. Plies:

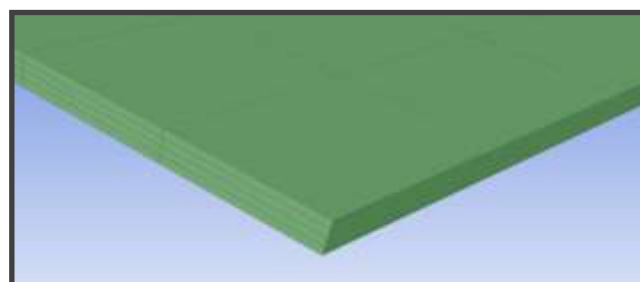


Figure5: Plies of specimen

IV. FINITE ELEMENT ANALYSIS

NormalStressPlot:

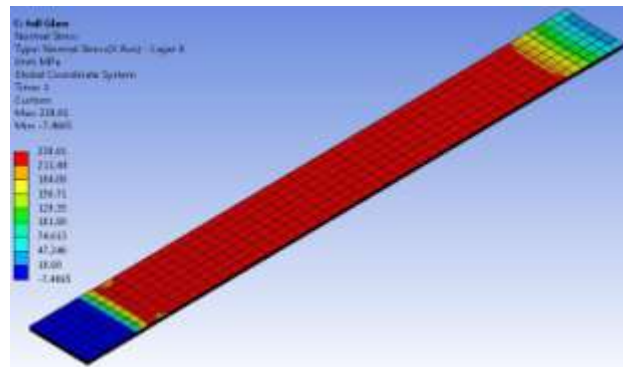


Figure7:NormalstressesofPureCompositeMaterial

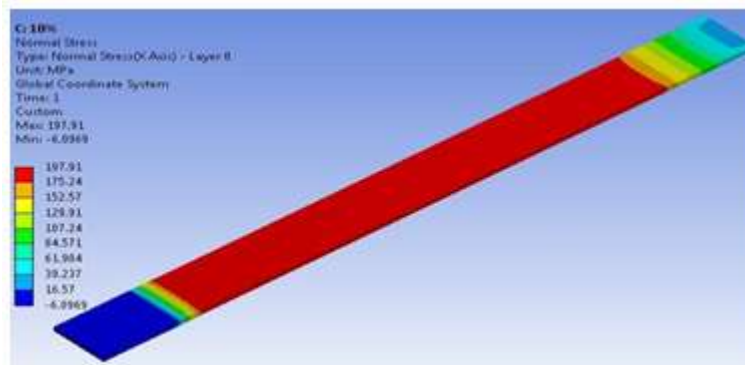


Figure 8: Normal stresses with the filler content of 10 % FlyashonCompositeMaterial

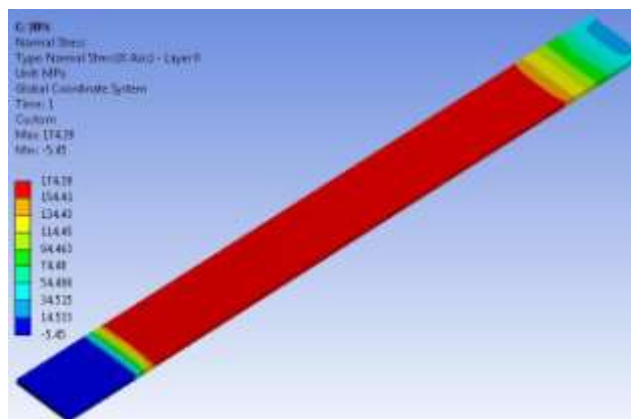


Figure 9: Normal stresses with the filler content of 30 % FlyashonCompositeMaterial

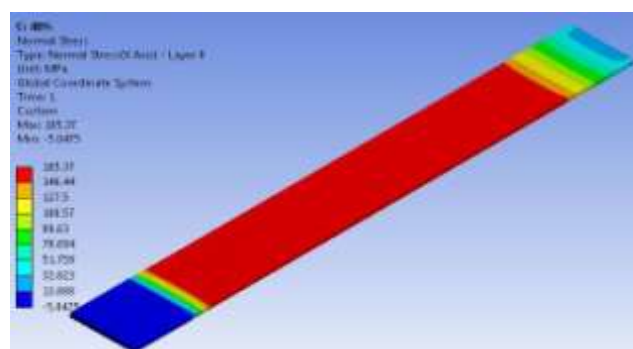


Figure 10: Normal stresses with the filler content of 40 %FlyashonCompositeMaterial

V. EXPERIMENTAL WORK

A. Fabrication Process Hand Lay-up Technique

The fabrication of composite material is done by Hand Lay-up technique. Glass mat is positioned manually in the open mould and resin is poured into glass plies. Entrapped air is removed manually with the roller to complete the lamin

ate structure.

The fibers are manually placed into one sided gel coated male or female mould. A matrix of thermosetting resin is rolled onto the fibers using hand roller. More layers can be added and, after drying, the composite part can be removed from the mould.



Figure 11: Test Specimens of various (10%, 30%, 40%) of flyash content and pure composite material.

A. Tensile Test of Composite Material:

The test specimens are fabricated in accordance with the ASTM D3039. The test is conducted on UTM/E-40 with a resolution of piston movement 0.1mm.



Figure 12: Specimen on UTM during tensile test



Figure13: Specimen of flyash content of 40% after testing

VI. RESULT AND DISCUSSION

Table3: Finite Element Analysis Results

Sr. No.	Filler content in % (flyash)	Force (N)	Normal Stress in (MPa)
1	-	10000	238.81
2	10	9500	197.91
3	30	9000	174.39
4	40	8500	165.37

Table4: Experimental Results

Sr. No.	Filler content in % (flyash)	Force (N)	Normal Stress in (MPa)
1	-	10789	213.33
2	10	10270	181.26
3	30	9751	159.95
4	40	8105	147.31

Above tables show the finite element analysis and experimental results of different composite specimens with the varying percentage of flyash and pure composite material.

VII. CONCLUSION

In this present work the residues from the thermal power plant is utilized as filler material in the glass fiber reinforced epoxy composites. The CAD model and analysis is carried out on ANSYS 16.0. The composite specimens are prepared on the basis of variation in filler content (10%, 30%, 40%) of fly ash and one pure composite material. Numerical analysis was performed on above three specimens and on one pure composite material. It is found that the pure composite material requires maximum forces (10000 N by FEM and 10789 N by Experimental) and composite material with 40% filler content of fly ash requires minimum forces (8500 N by FEM and 8105 N by Experimental). Thus by increasing the percentage of filler content the tensile strength is decreases. It is concluded that pure composite material showed significant strength when compare to filler content of 10%, 30% and 40%.

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